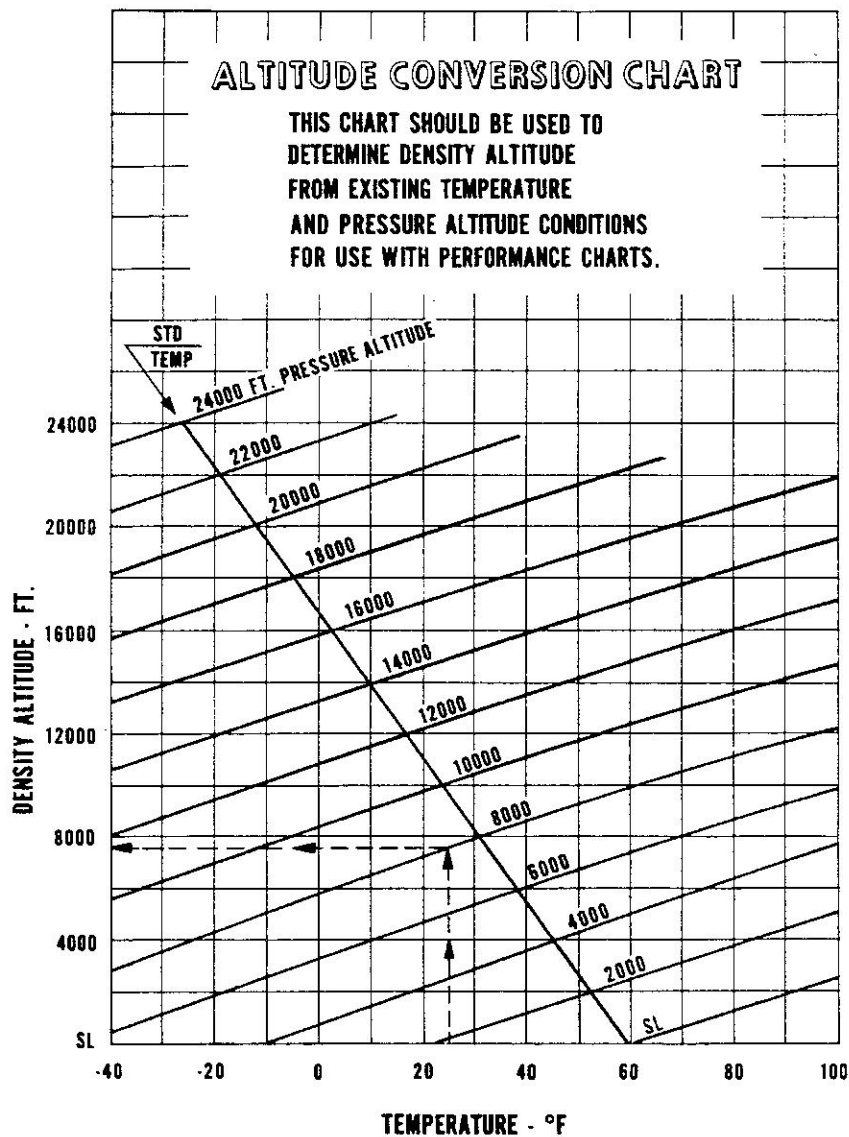


PERFORMANCE CHARTS

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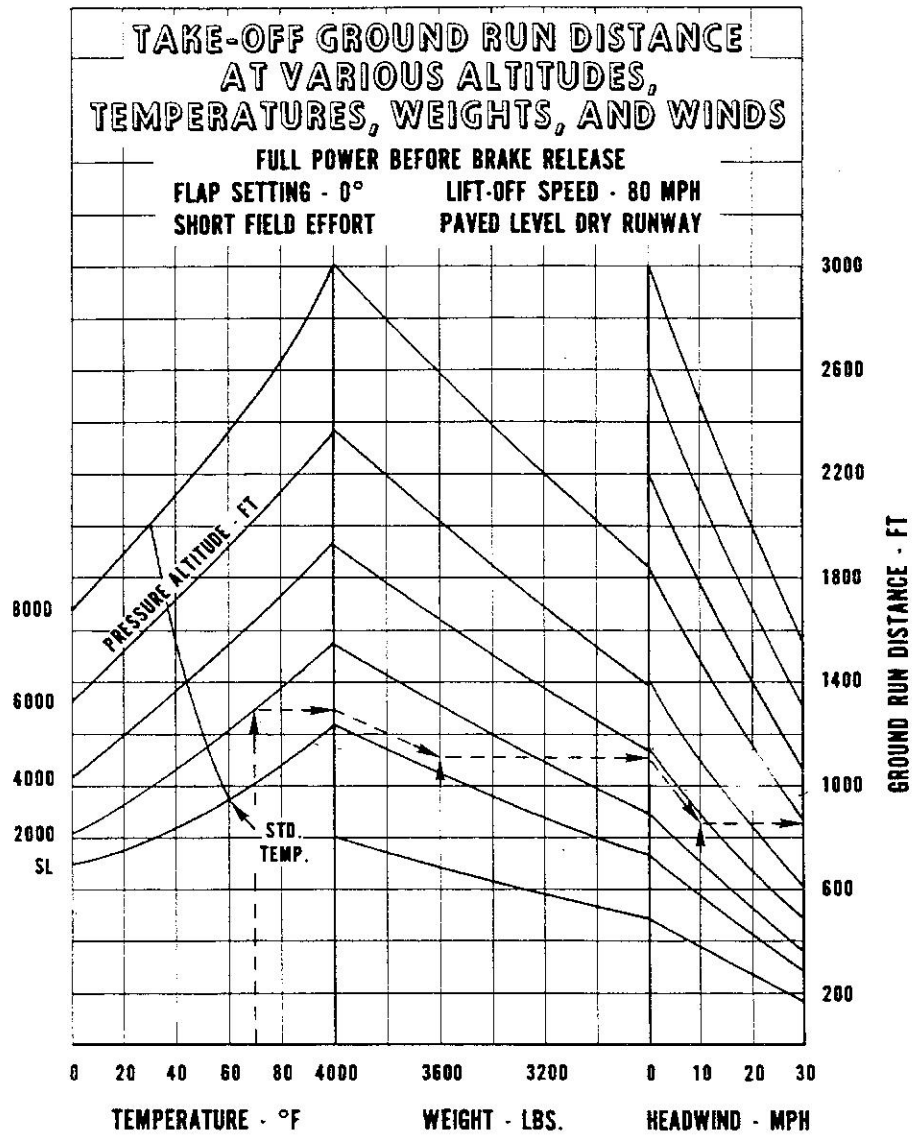
WARNING

Performance information derived by extrapolation beyond the limits shown on the charts should not be used for flight planning purposes.



Example: Temp. 25° F
Density Alt. 7500 Ft.

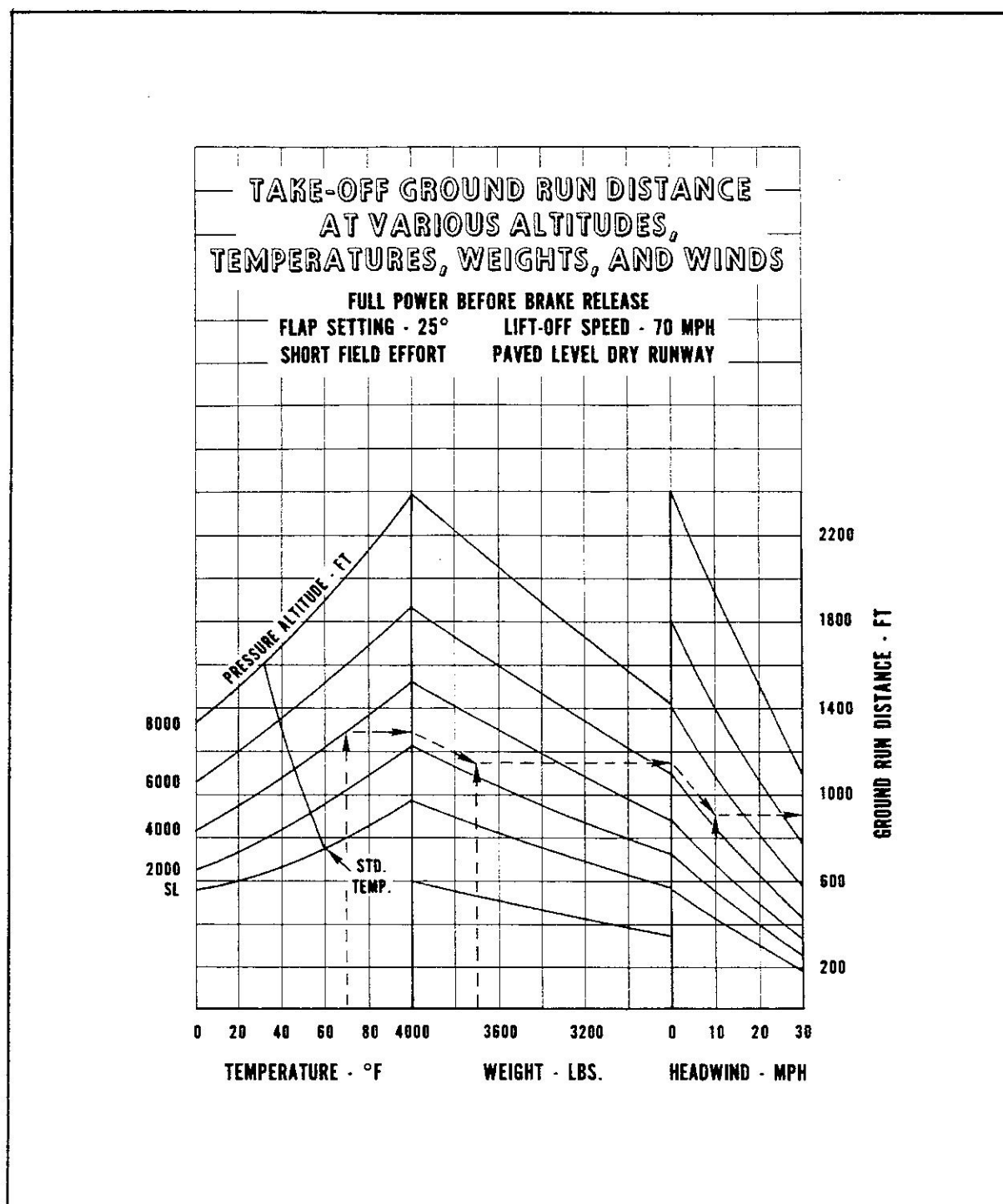
Press. Alt. 8000 Ft.



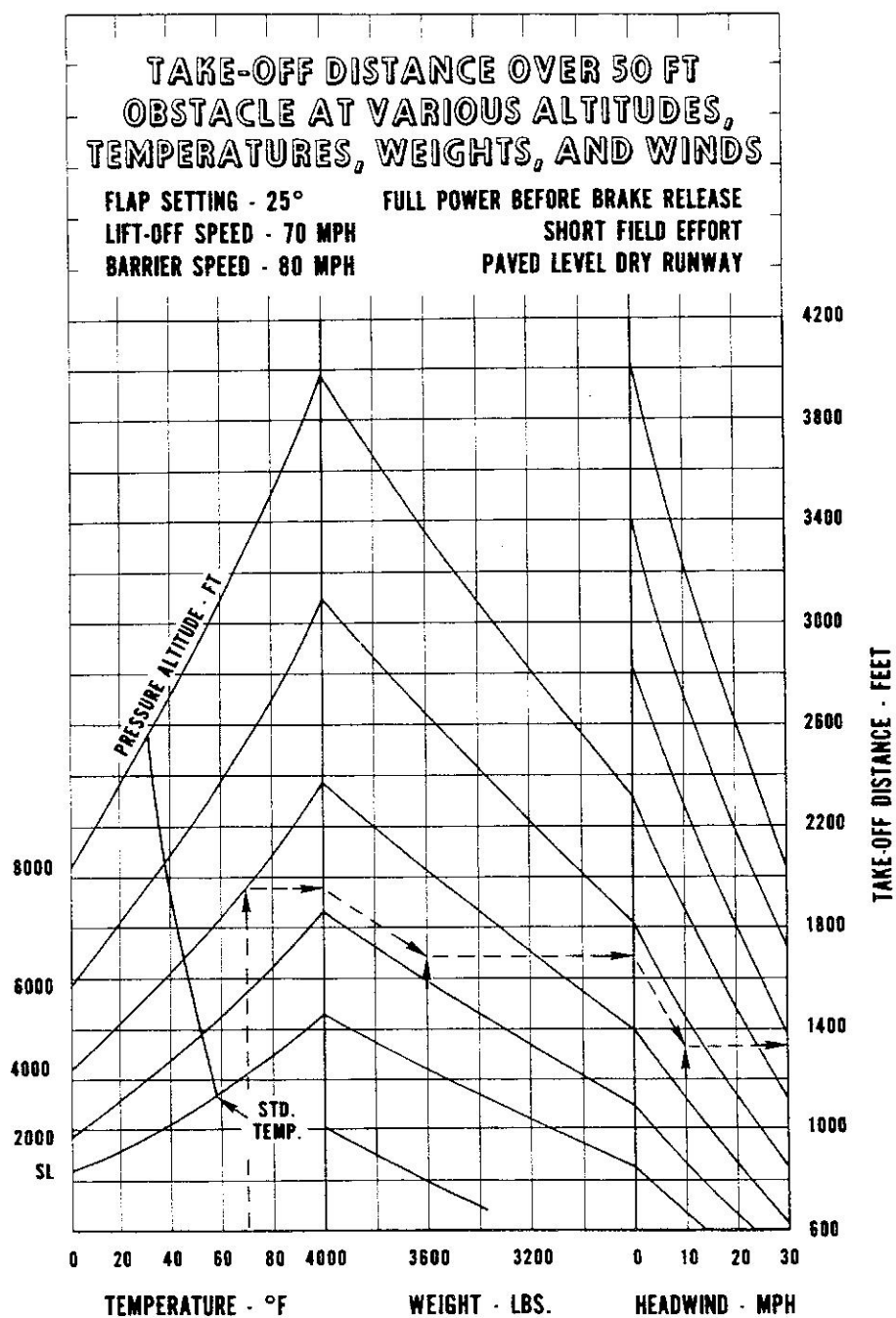
Example: Temp. 70° F
Press. Alt. 2000 Ft.

Wt. 3600 lbs.
Hd. wind 10 MPH

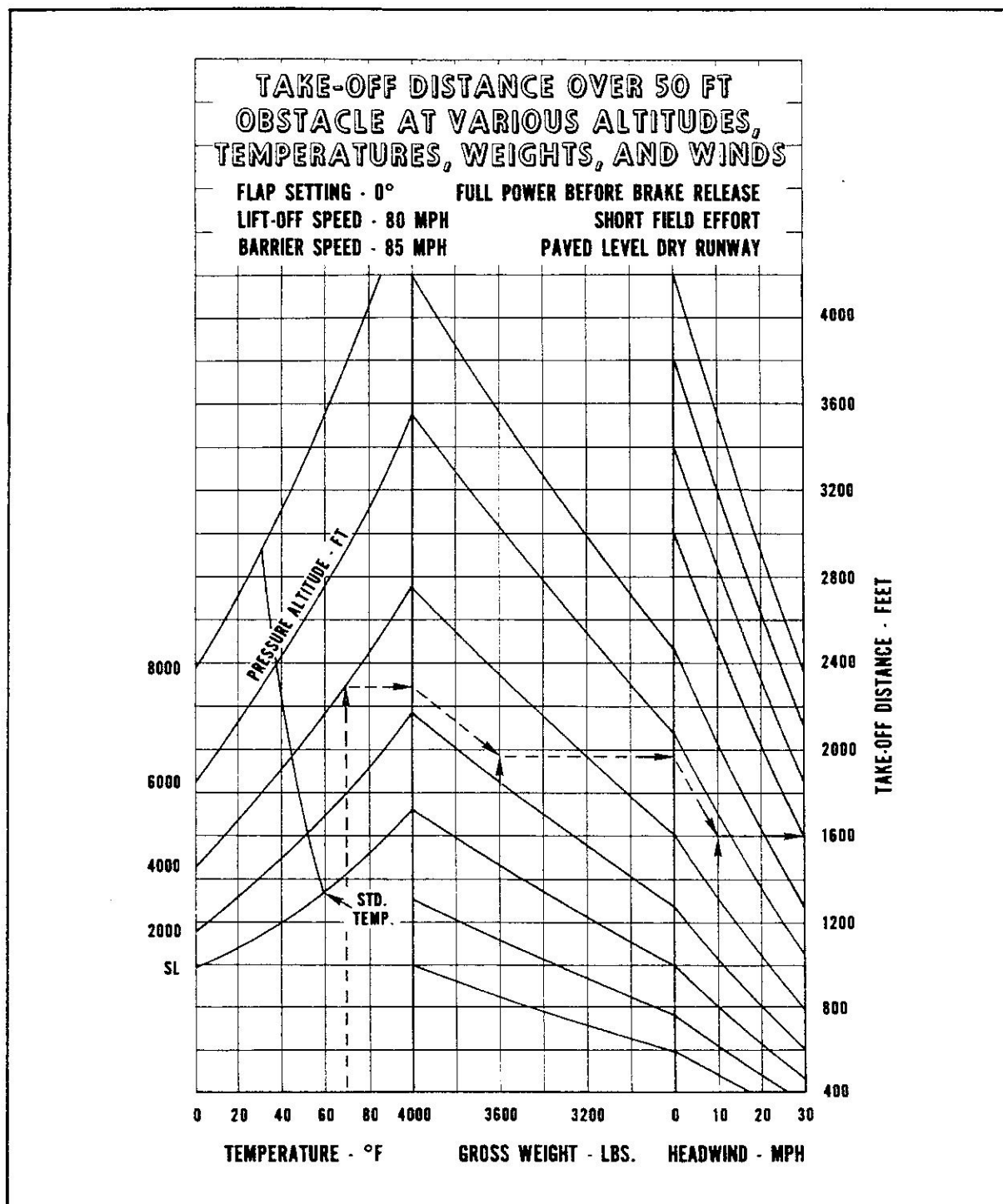
Ground run 850 Ft.



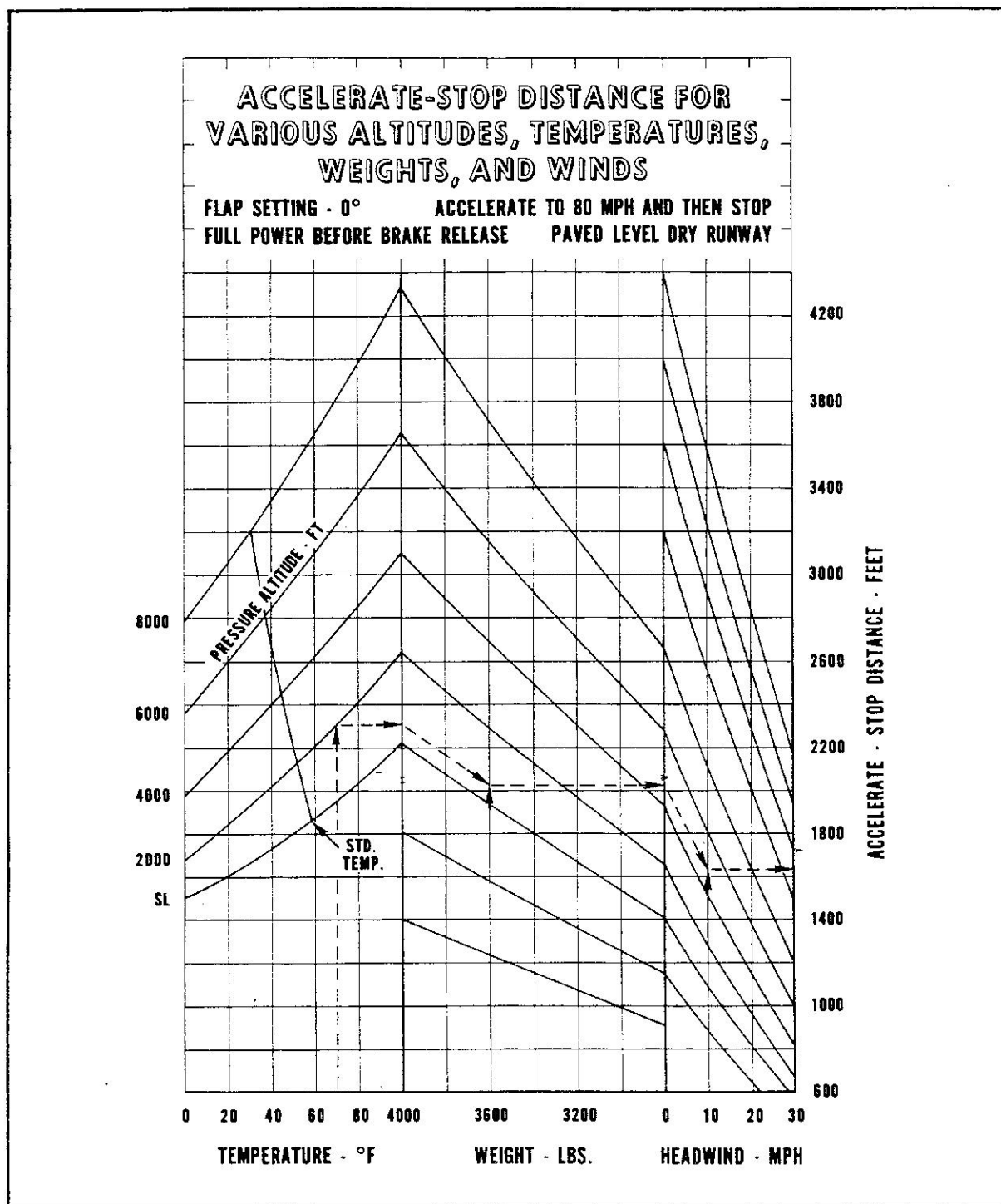
Example: Temp. 70° F Wt. 3700 lbs Ground Run 900 ft
 Press. Alt. 4000 ft Hd. wind 10 MPH



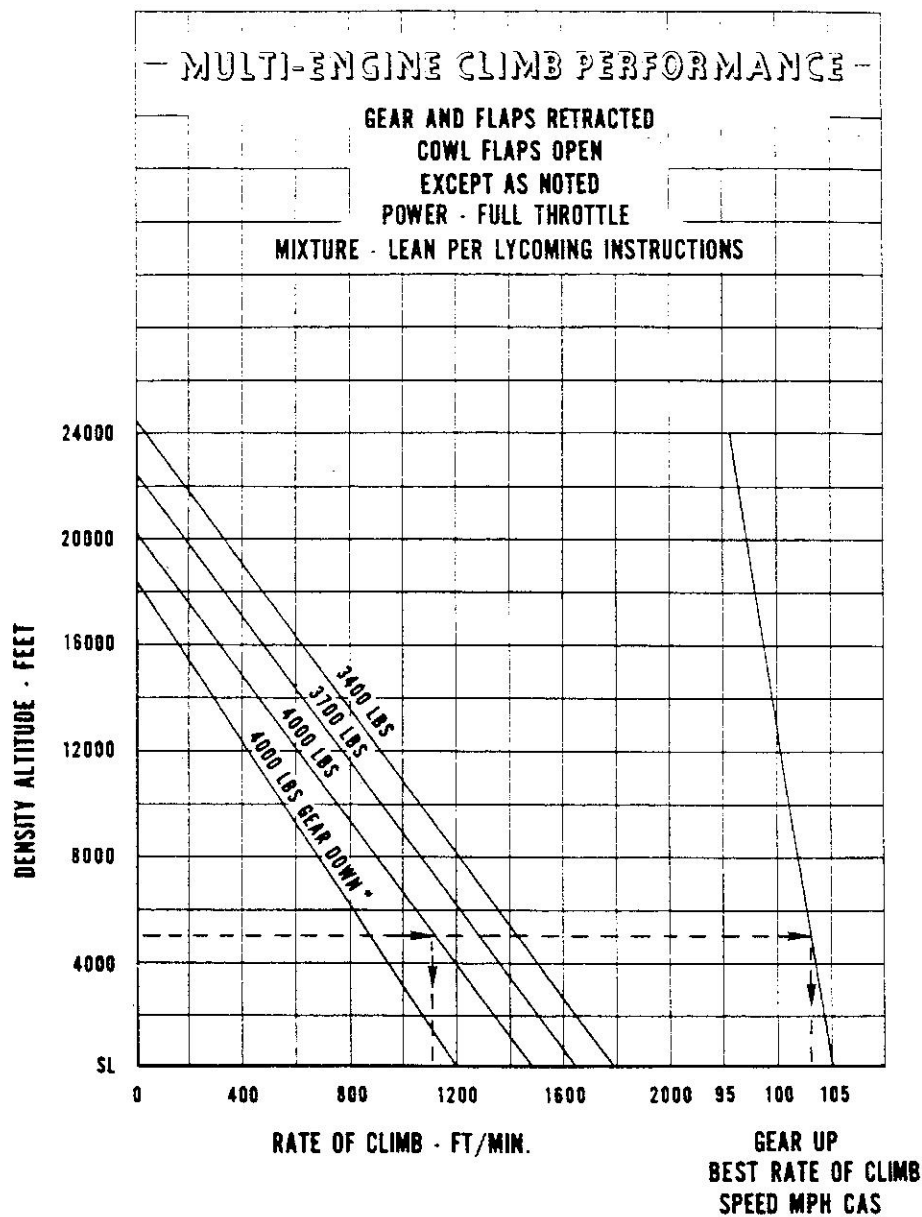
Example: Temp. 70° F Wt. 3600 lbs T. O. Dist. 1350 ft
Press. Alt. 4000 ft Hd. wind 10 MPH



Example: Temp. 70° F Wt. 3600 lbs T. O. Dist. 1600 ft
 Press. Alt. 4000 ft Hd. wind 10 MPH

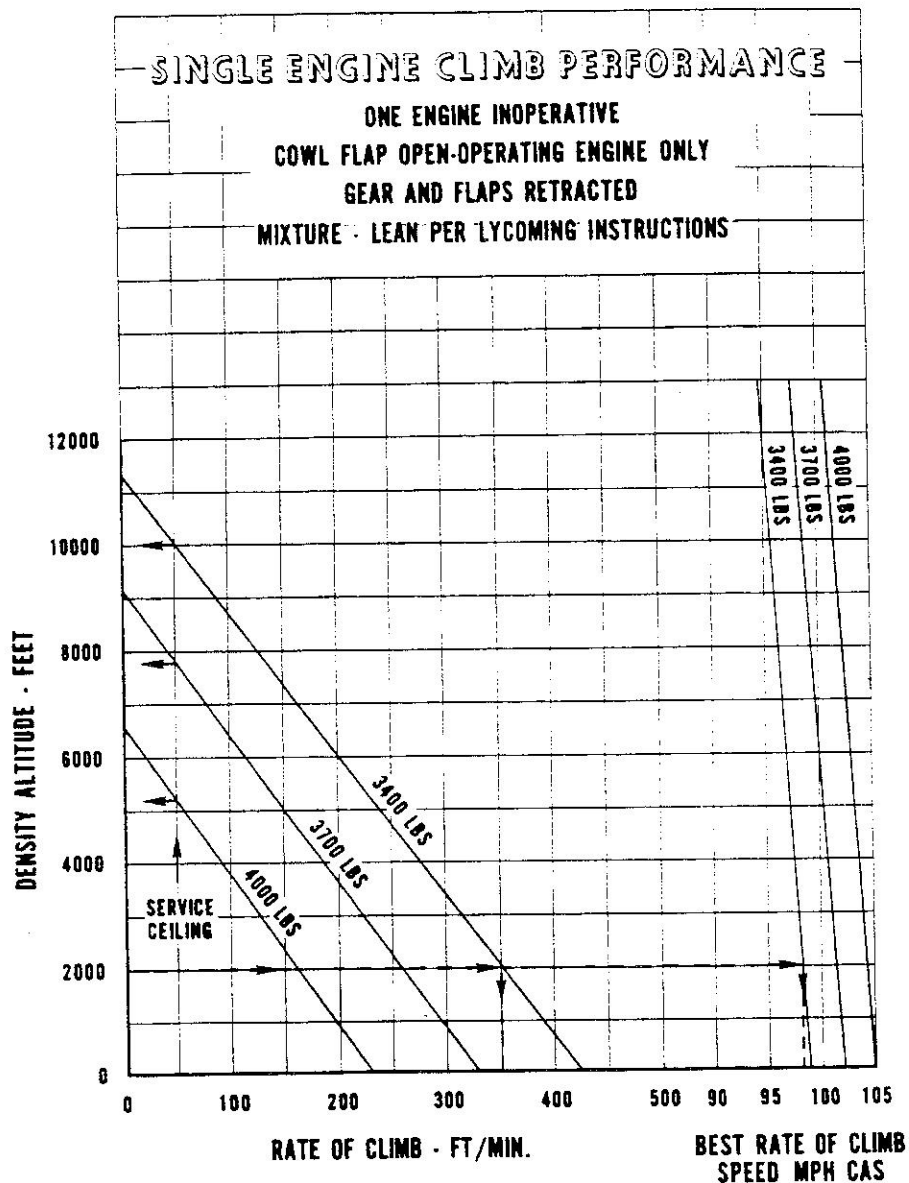


Example: Temp. 70° F Wt. 3600 lbs Accel. - Stop Dist. 1630 ft
 Press. Alt. 2000 ft Hd. wind 10 MPH

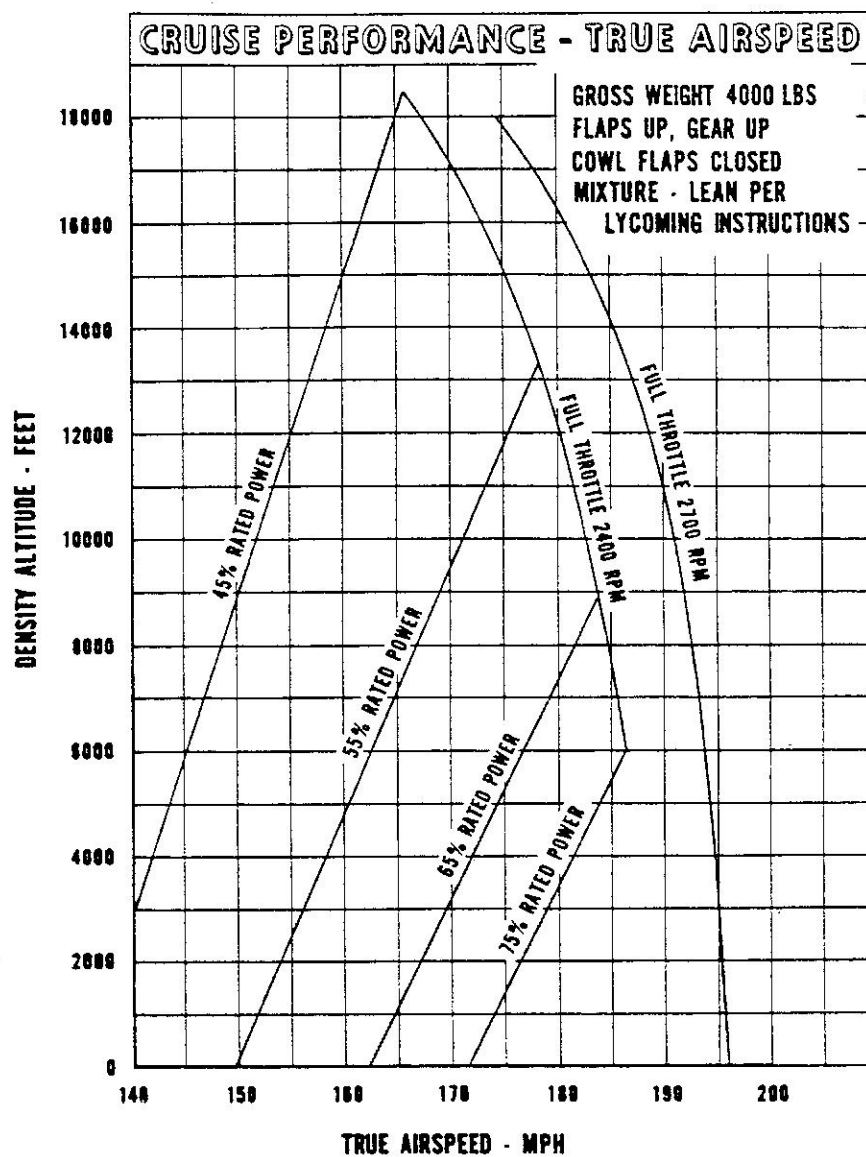


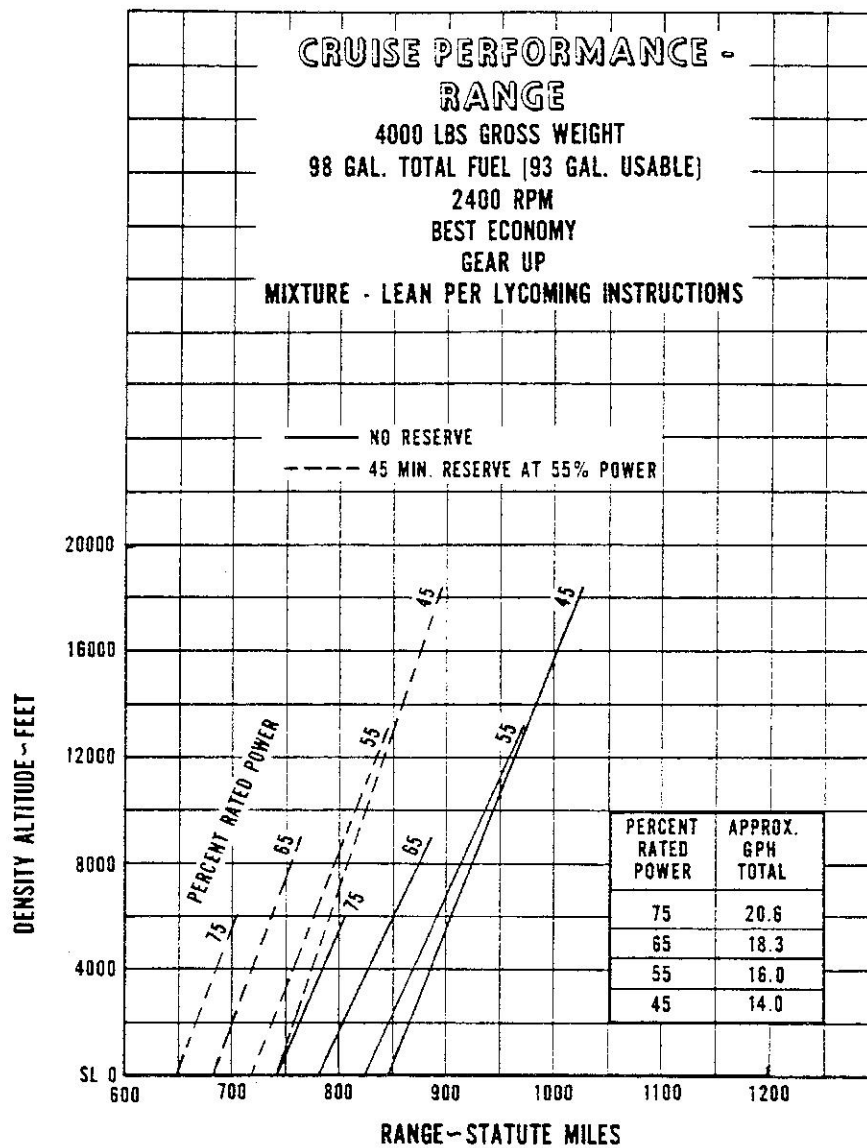
*Gear down best rate of climb speed is 92 MPH CAS at all altitudes.

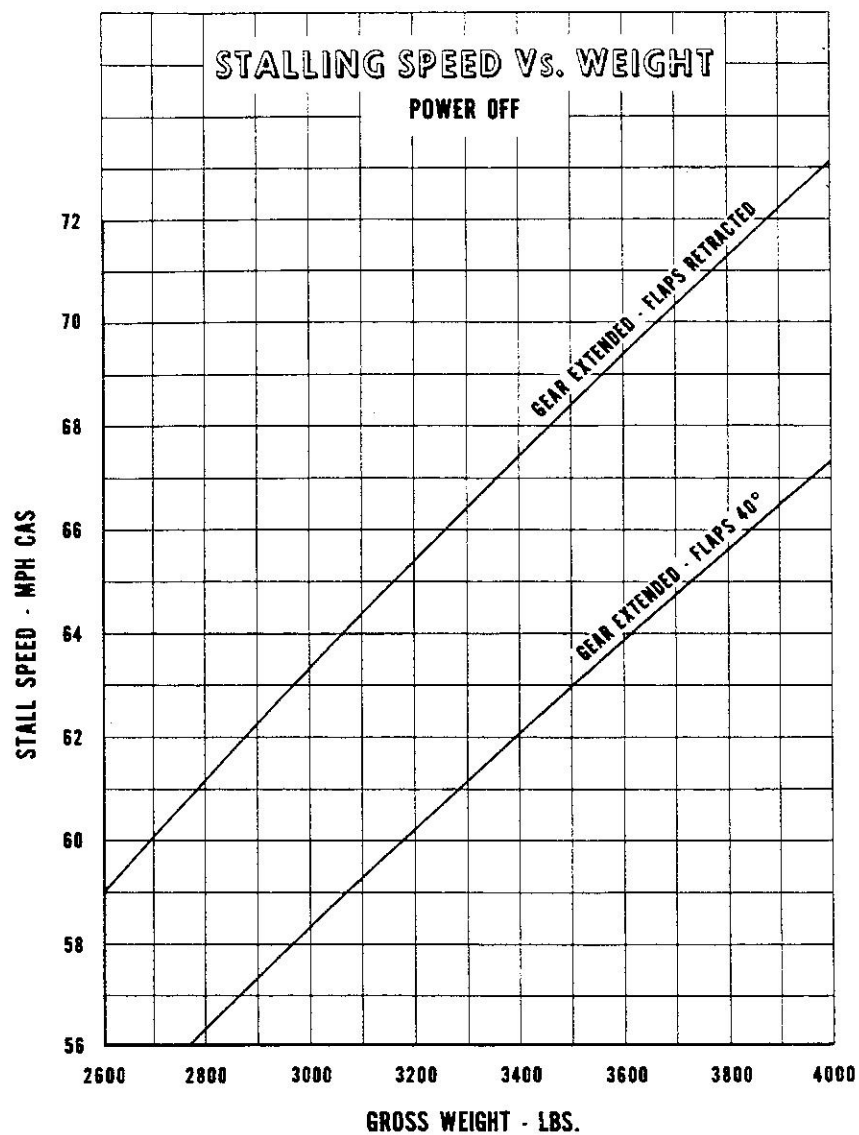
Example: Wt. 4000 lbs Rate of Climb 1100 ft/min
 Den. Alt. 5000 ft Best R/C Speed 103 MPH
 (Gear Up)

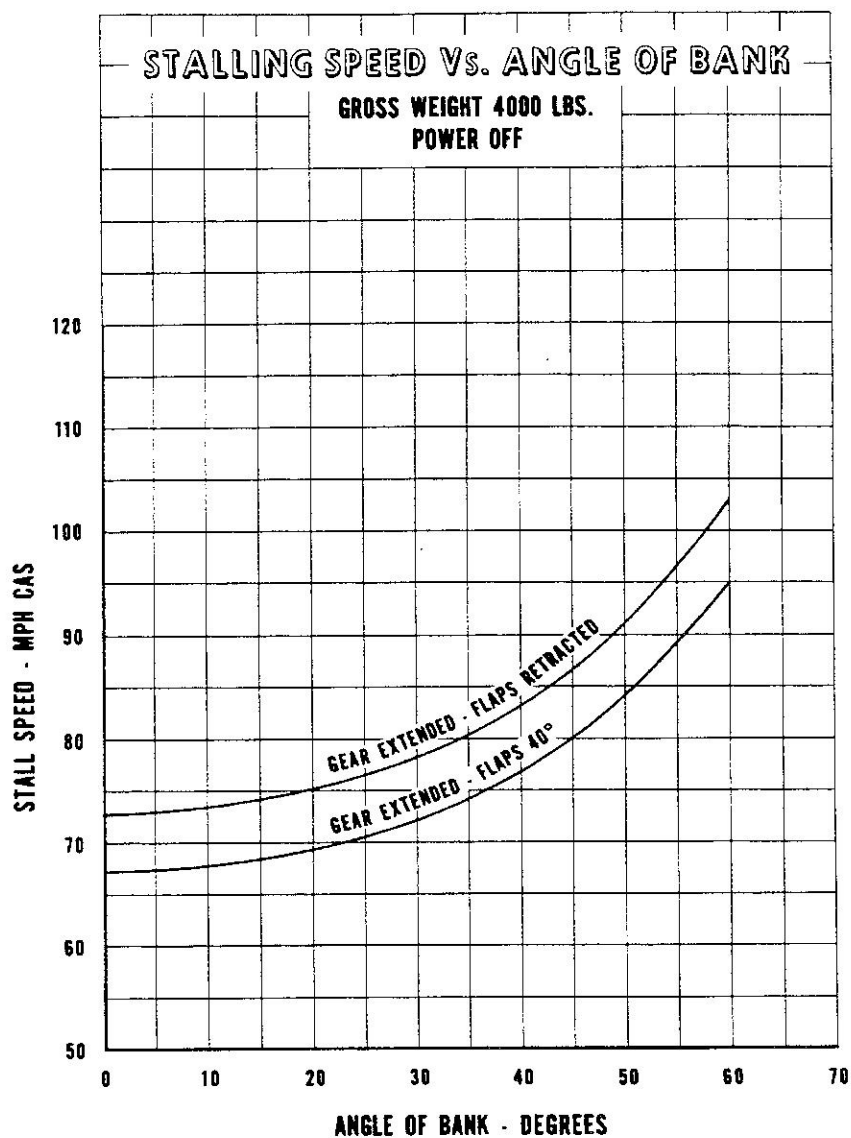


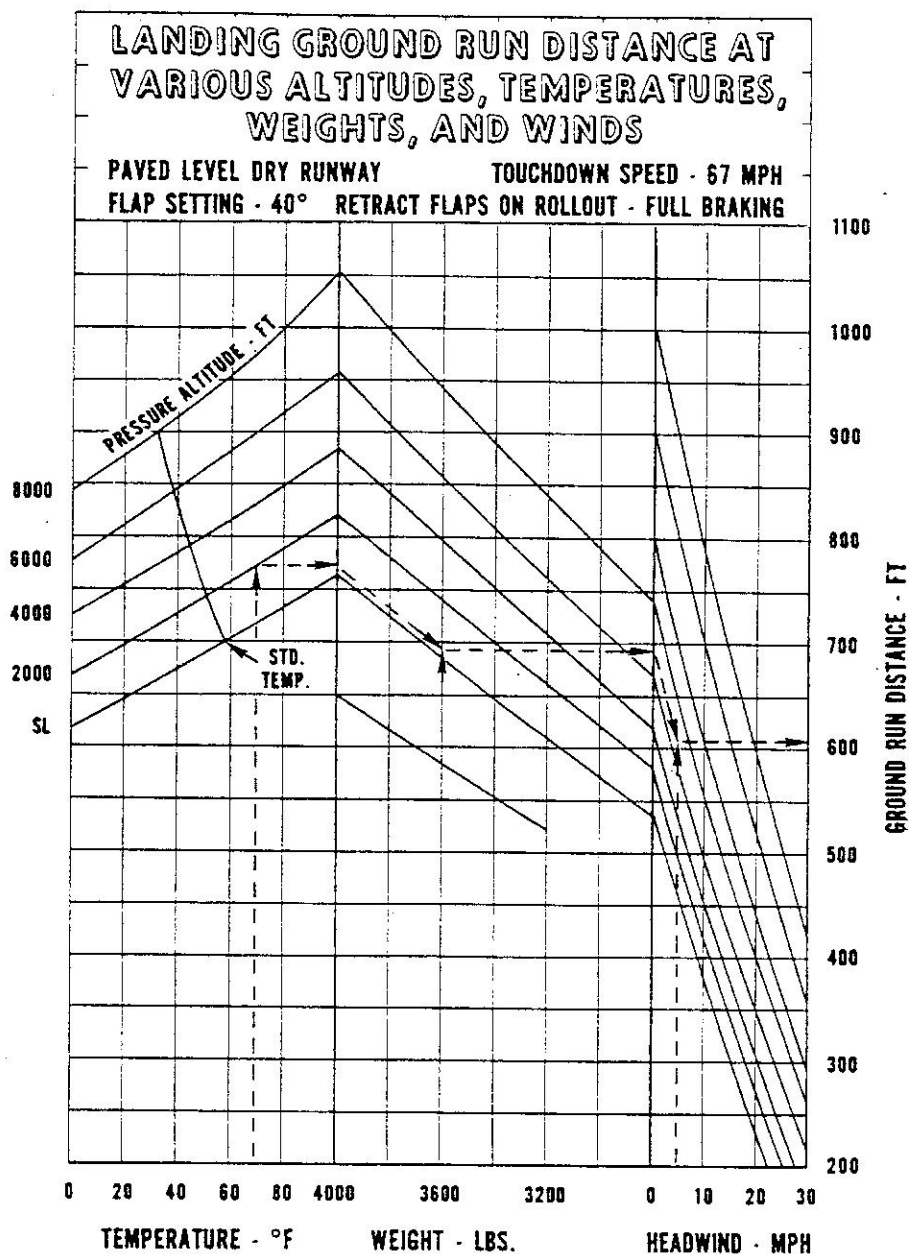
Example: Wt. 3400 lbs Rate of Climb 350 ft/min
 Den. Alt. 2000 ft Best R/C Speed 98 MPH





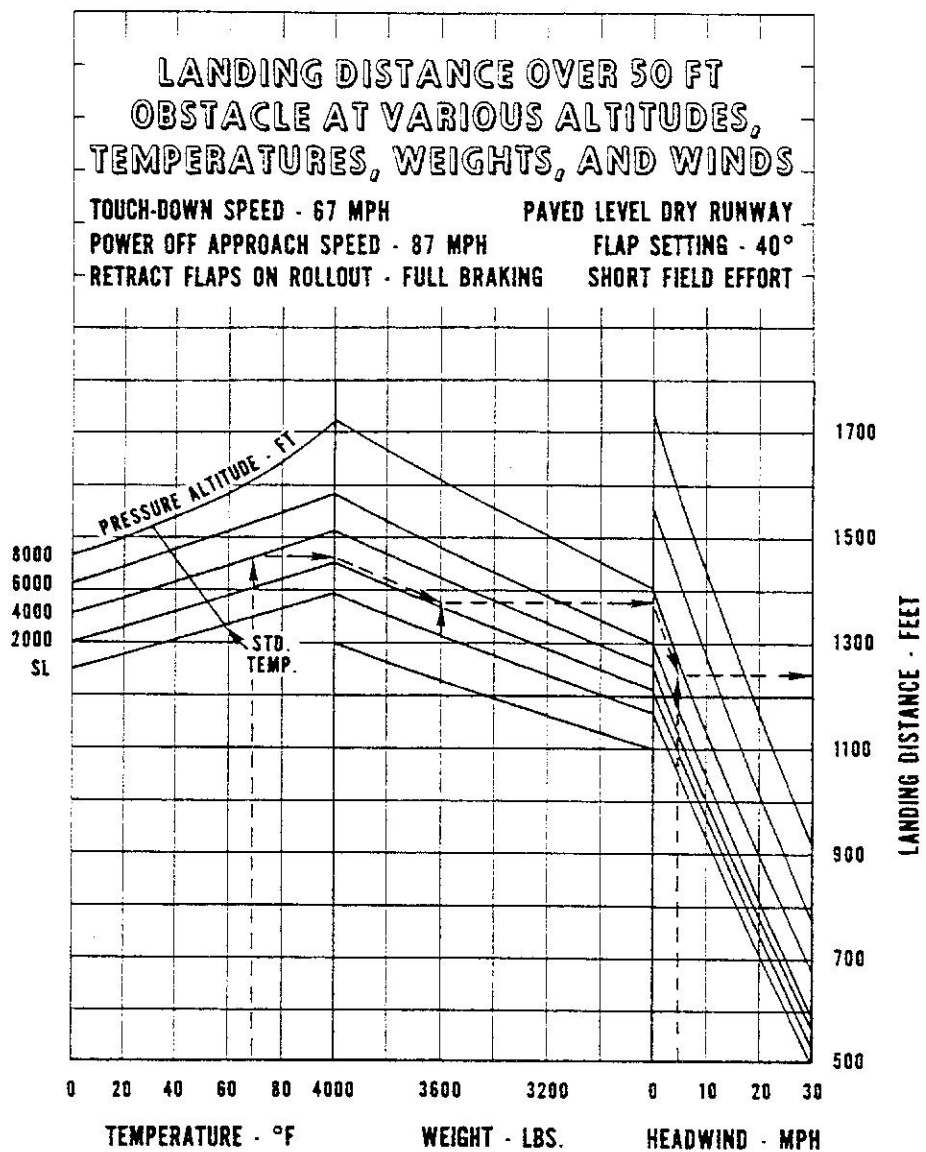






NOTE: The above distances may be reduced by approximately 25% when the aircraft is equipped with Heavy Duty Wheels, Tires and Brakes. (Reference Aircraft Equipment List in Weight and Balance Section of this Manual.)

Example: Temp. 70° F Wt. 3600 lbs Ground Run 615 ft
Press. Alt. 2000 ft Hd. wind 5 MPH



NOTE: The above distances may be reduced by approximately 12% when the aircraft is equipped with Heavy Duty Wheels, Tires and Brakes. (Reference Aircraft Equipment List in Weight and Balance Section of this Manual.)

Example: Temp. 70° F Wt. 3600 lbs. Landing Dist. 1240 ft
Press. Alt. 4000 ft Hd. wind 5 MPH

Power Setting Table - Lycoming Model IO-360-C Series, 200 HP Engine

Press. Alt Feet	Std. Alt Temp ° F	110 HP - 55% Rated RPM AND MAN. PRESS.		130 HP - 65% Rated RPM AND MAN. PRESS.		150 HP - 75% Rated RPM AND MAN. PRESS.		Press. Alt Feet
		2100	2400	2100	2400	2100	2400	
SL	59	22.9	20.4	25.9	22.9	25.5	25.5	SL
1,000	55	22.7	20.2	25.6	22.7	25.2	25.2	1,000
2,000	52	22.4	20.0	25.4	22.5	25.0	25.0	2,000
3,000	48	22.2	19.8	25.1	22.2	24.7	24.7	3,000
4,000	45	21.9	19.5	24.8	22.0	24.4	24.4	4,000
5,000	41	21.7	19.3	FT	21.7	FT	FT	5,000
6,000	38	21.4	19.1	--	21.5	--	--	6,000
7,000	34	21.2	18.9	--	21.3	--	--	7,000
8,000	31	21.0	18.7	--	21.0	--	--	8,000
9,000	27	FT	18.5	--	FT	--	--	9,000
10,000	23	--	18.3	--	--	--	--	10,000
11,000	19	--	18.1	--	--	--	--	11,000
12,000	16	--	17.8	--	--	--	--	12,000
13,000	12	--	17.6	--	--	--	--	13,000
14,000	9	--	FT	--	--	--	--	14,000

To maintain constant power, correct manifold pressure approximately 0.16" Hg for each 10° F variation in inlet air temperature from standard altitude temperature. Add manifold pressure for air temperatures above standard; subtract for temperatures below standard.