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NORMAL PROCEDURES

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SECTION 4

NORMAL PROCEDURES

4.1 GENERAL

This section provides the normal operating procedures for the PA-44-180, Seminole airplane. All normal operating procedures required by FAA regulation, as well as those deemed necessary for normal operation of the airplane are presented.

This section provides checklists for all normal operating procedures, using a simple action - reaction format, with little emphasis on system operation. These checklists should be used during normal ground and flight operations.

When appropriate, additional information is provided immediately below the checklist, providing more detailed information related to that procedure. In order to operate the airplane in a safe and efficient manner, pilots should familiarize themselves with the both the checklists and amplified procedures.

Normal operating procedures associated with optional systems and equipment, which require handbook supplements, are presented in Section 9, Supplements.

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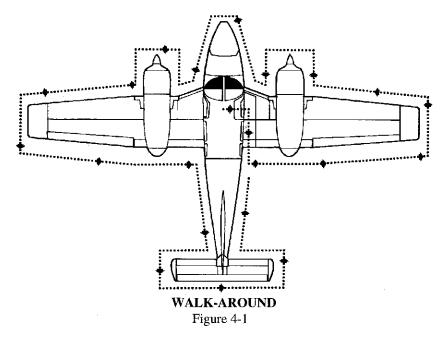
4.3 AIRSPEEDS FOR SAFE OPERATIONS

The following airspeeds are for standard airplanes flown at gross weight under standard conditions at sea level.

Performance for a specific airplane may vary from published figures depending upon the equipment installed, the condition of the engine, airplane and equipment, atmospheric conditions and piloting technique.

(a) Best Rate of Climb Speed (Vy)
(b) Best Angle of Climb Speed (Vx)
(c) Maximum Operating Maneuvering
Speed (Vo)135 KIAS (115 KIAS @ 2870 LB)
See Airspeed Limitations Section 2.3
(d) Maximum Flap Speed
(e) Landing Final Approach Speed (Flaps 40 degrees)
Short Field Effort75 KIAS
(f) Intentional One Engine Inoperative Speed (VSSE) 82 KIAS
(g) Maximum Demonstrated Crosswind Velocity 17 KIAS

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4.5 NORMAL PROCEDURES CHECKLIST

The airplane should be given a thorough preflight and walk-around check. The preflight should include a check of the airplane's operational status, computation of weight and C.G. limits, takeoff distance and in-flight performance. A weather briefing should be obtained for the intended flight path, and any other factors relating to a safe flight should be checked before takeoff.

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4.5 NORMAL PROCEDURES CHECKLIST (continued)

4.5a Preflight Checklists

COCKPIT

Control Wheel	release restraints
Flight Controls	
PARK BRAKE	
Static System	DRAIN
Alternate Static Source	
LEFT/RIGHT ENG MAG Switches	OFF
LEFT/RIGHT FUEL PUMP Switches.	OFF
GEAR	DOWN
THROTTLES	IDLE
MIXTURES	CUT-OFF
COWL FLAPS	OPEN
Stabilator & Rudder Trim	
FUEL Selectors	
All Electrical Switches	OFF
BATT MASTR Switch	ON
FUEL QTY Indications	CHECK
Gear Position Indications	3 GREEN
BATT MASTR Switch	OFF
Emergency Exit	CHECK
FLAPS	EXTEND
Windows	check CLEAN
Required Papers	
POH	
Baggage	STOW PROPERLY - SECURE

CAUTION

If the emergency exit is unlatched in flight, it may separate and damage the exterior of the airplane.

Set the parking brake by first depressing and holding the toe brake pedals and then pulling out the PARK BRAKE knob. The static drains are located on the lower left sidewall adjacent to the pilot.

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4.5 NORMAL PROCEDURES CHECKLIST (continued)

4.5a Preflight Checklists (continued)

RIGHT WING

Fuel Sump Drains	DRAIN
Surface Condition	
Flap and Hinges	
Aileron, Hinges & Freedom of Movement	
Static Wicks	CHECK
Wing Tip and Lights	
Scupper Drain	
Fuel Tank Vent	
Tie Down	
Nacelle Fuel Filler Cap	CHECK & SECURE
Engine Oil & Cap	CHECK & SECURE
Propeller & Spinner	CHECK
Air Inlets	CLEAR
Cowl Flap Area	CHECK
Main Gear Strut	
	$(2.60 \pm 0.25 \text{ in.})$
Main Wheel Tire	CHECK
Brake, Block & Disc	CHECK
Chock	REMOVE

Eight quarts of oil are required for maximum range flights.

NOSE SECTION

4.5a Preflight Checklists (continued)

NOSE SECTION	
	CHECK
	CLEAN
	CLEAR
Landing Lights	CHECK
Chock	REMOVE
Nose Gear Strut	PROPER INFLATION
	(2.70 + - 0.25 in.)
Nose Wheel Tire	CHECK
LEFT WING	
	CLEAR of ICE, FROST & SNOW
Main Gear Strut	PROPER INFLATION
	(2.60 +/- 0.25 in.)
Main Wheel Tire	(2.60 +/- 0.25 in.)
Main Wheel Tire	(2.60 +/- 0.25 in.)
Main Wheel TireBrake, Block & Disc	(2.60 +/- 0.25 in.)
Main Wheel Tire Brake, Block & Disc Chock	(2.60 +/- 0.25 in.)
Main Wheel Tire	(2.60 +/- 0.25 in.)
Main Wheel Tire Brake, Block & Disc Chock Cowl Flap Area Nacelle Fuel Filler Cap	(2.60 +/- 0.25 in.) ————————————————————————————————————
Main Wheel Tire Brake, Block & Disc Chock Cowl Flap Area Nacelle Fuel Filler Cap Engine Oil & Cap	(2.60 +/- 0.25 in.) CHECK CHECK REMOVE CHECK CHECK & SECURE CHECK & SECURE
Main Wheel Tire Brake, Block & Disc Chock Cowl Flap Area Nacelle Fuel Filler Cap Engine Oil & Cap Propeller & Spinner	(2.60 +/- 0.25 in.) CHECK CHECK REMOVE CHECK CHECK & SECURE CHECK & SECURE CHECK & CHECK
Main Wheel Tire Brake, Block & Disc Chock Cowl Flap Area Nacelle Fuel Filler Cap Engine Oil & Cap Propeller & Spinner Air Inlets	(2.60 +/- 0.25 in.) CHECK CHECK REMOVE CHECK & SECURE CHECK & SECURE CHECK CHECK CHECK CHECK CHECK
Main Wheel Tire Brake, Block & Disc Chock Cowl Flap Area Nacelle Fuel Filler Cap Engine Oil & Cap Propeller & Spinner Air Inlets Scupper Drain	(2.60 +/- 0.25 in.)
Main Wheel Tire Brake, Block & Disc Chock Cowl Flap Area Nacelle Fuel Filler Cap Engine Oil & Cap Propeller & Spinner Air Inlets Scupper Drain Fuel Tank Vent	(2.60 +/- 0.25 in.) CHECK CHECK REMOVE CHECK & SECURE CHECK & SECURE CHECK CHECK CHECK CHECK CHECK

Stall Warning Vanes CHECK
Pitot/ Static Head CLÉAR
Wing Tip and Lights CHECK
Aileron, Hinges & Freedom of Movement CHECK
Flap and Hinges CHECK
Static Wicks CHECK

Eight quarts of oil are required for maximum range flights.

4.5a Preflight Checklists (continued)
FUSELAGE (LEFT SIDE)
General Condition CHECK Emergency Exit CHECK Antennas CHECK Fresh Air Inlet CLEAR
EMPENNAGE
Surface Condition
Stabilator, Trim Tab & Freedom of Movement
Rudder, Trim Tab & Freedom of Movement
Static Wicks
FUSELAGE (RIGHT SIDE)
General Condition CHECK
Baggage DoorCLOSED AND LATCHED
Cabin Door
MISCELLANEOUS
FLAPSRETRACT
BATT MASTR SwitchON
Interior Lighting (Night Flight)ON & CHECK
CAUTION CAUTION
Care should be taken when checking the heated
pitot head. The unit becomes very hot. Ground
operation should be limited to 3 minutes maximum

PITOT HEAT Switch ON
Exterior Lighting Switches ON & CHECK
Pitot/Static Head CHECK - WARM
PITOT HEAT Switch OFF
All Lighting Switches OFF
BATT MASTR Switch OFF
Passengers BOARD

to avoid damaging the heating elements.

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4.5b Before Starting Engine Checklists

BEFORE STARTING ENGINE

Preflight Check	COMPLETED	
Flight Planning		
Cabin DoorCLOSE & LA	TCH (Lower then Upper)	
Seats		
Seatbelts and Harness		
	CHECK INERTIA REEL	
PARK BRAKE	SET	
GEAR		
THROTTLES	IDLE	
PROPELLERS		
MIXTURES	CUT-OFF	
Friction Handle	AS DESIRED	
ALT-AIR	CLOSE	
COWL FLAPS	OPEN	
Stabilator & Rudder Trim	SET	
FUEL Selectors	ON	
LEFT/RIGHT ALTR SwitchesON (OFF, if external power connected)		
EMERG BATT		
E VOLTS Indication	. 23.3 VOLTS (minimum)	
AVION MASTER Switch	OFF	
STROBE LIGHTS	FIN STROBE	
All Other Electrical Switches	OFF	
CABIN HEAT Switch	OFF	
Circuit Breakers	CHECK	
Proceed with appropriate Engine Start Checklis	t.	

NOTE

The EMERG BATT should remain ON after checking for proper bus operation, allowing the PFD to remain powered for engine start. Avoid delays between this check and engine starting to preserve emergency battery power.

If the E VOLTS indication is less than 23.3 VOLTS, the voltage should be checked again at the end of the GROUND CHECK checklist (after being charged for some time by the primary electrical system). If E VOLTS is still less than 23.3 volts, determine the cause and correct the issue prior to flight.

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4.5c Engine Start Checklists

ENGINE START - GENERAL

WARNING

The L START ENGD or R START ENGD warning CAS message will illuminate after 30 seconds of continuous engine cranking. If the CAS message illuminates after the engine is running, stop the engine and determine the cause.

NOTE

When starting at ambient temperatures +20°F and below, operate first engine started with alternator ON (at max charging rate not to exceed 1500 RPM) for 5 minutes minimum before initiating start on second engine.

NOTE

Starter manufacturer recommends starter cranking periods be limited to 10 seconds with a 20 second rest period between cranking attempts. Maximum of 6 start periods allowed. If a start is not achieved on sixth attempt allow starter to cool for 30 minutes before attempting additional starts. Do not engage the starter immediately after releasing it. This practice may damage the starter mechanism.

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NORMAL START - COLD ENGINE

	ON	
Gear Position Indications	3 GREEN	
	CONSIDER ANY ILLUMINATED	
PFD Annunciations	CONSIDER ANY ILLUMINATED	
THROTTLES	1/4 inch OPEN	
PROPELLERS	FULL INCREASE	
*FUEL PUMP	ON	
*MAG LEFT/RIGHT Switches	ON	
*MIXTURE	PRIME - then CUT-OFF	
*Propeller Area	CLEAR	
	ENGAGE	
	ADVANCE	
	ADJUST WHEN ENGINE	
	STARTS TO 1000 RPM	
*OIL PSI	CHECK	
Repeat above procedure (*) for second engine start		
VOLTS	CHECK	
	CHECK	
	OFF	

When the engine starts, adjust the throttle and monitor the oil pressure. If no oil pressure is indicated within 30 seconds, shut down the engine and have it checked. In cold weather it may take somewhat longer for an oil pressure indication.

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NORMAL START - HOT ENGINE

BATT MASTR Switch	ON	
Gear Position Indications	3 GREEN	
CAS Messages	. CONSIDER ANY ILLUMINATED	
PFD Annunciations	. CONSIDER ANY ILLUMINATED	
THROTTLES		
PROPELLERS	FULL INCREASE	
	CUT-OFF	
*FUEL PUMP	ON	
*MAG LEFT/RIGHT Switches	ON	
	CLEAR	
	ENGAGE	
*MIXTURE	ADVANCE	
	ADJUST to LOW RPM	
	CHECK	
If engine does not start, add small amount of prime and repeat above.		
Repeat above procedure (*) for seco	and engine start	
	CHECK	
	CHECK	
FUEL PUMPS		

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ENGINE START - COLD WEATHER (BELOW 10°F)

WARNING

Ensure all magneto (MAG LEFT/RIGHT) and master switches (BATT MASTR) are OFF and mixture controls are in cut-off before turning propeller manually.

If available, preheat should be considered. Rotate each propeller through 10 blades manually during preflight inspection.

DATE MACED Coult-1	OFF
BATT MASTR SWITCH	OFF
LEFT/RIGHT ALTR Switches	VERIFY OFF
External Power (24-28 VDC source)CONNECT
Gear Position Indications	3 GREEN
CAS Messages	. CONSIDER ANY ILLUMINATED
PFD Annunciations	. CONSIDER ANY ILLUMINATED
	1/4 inch OPEN
PROPELLERS	FULL INCREASE
*FUEL PUMP	ON
*MAG LEFT/RIGHT Switches	ON
	PRIME - then CUT-OFF
*Propeller Area	CLEAR
*ENG START	ENGAGE
*MIXTURE	ADVANCE
*OIL PSI	CHECK

If engine does not start, add prime and repeat above.

Repeat above procedure (*) for second engine start	
THROTTLES	LOWEST POSSIBLE RPM
BATT MASTR Switch	ON
External Power	DISCONNECT
LEFT/RIGHT ALTR Switches	ON
VOLTS	CHECK
ALTR AMPS	CHECK
FUEL PUMPS	OFF

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ENGINE START - COLD WEATHER (BELOW 10°F) (continued)

After engine start and the throttle is set as desired, the oil pressure should be checked for a positive indication. If no oil pressure is indicated within 30-seconds, shut down the engine and have it checked. In cold weather it may take somewhat longer for an oil pressure indication.

NOTÉ

When starting at ambient temperatures +20°F and below, operate first engine started with alternator ON (not to exceed 1500 RPM) for 5 minutes minimum before initiating start on second engine.

ENGINE START WHEN FLOODED

BATT MASTR Switch	ON	
	3 GREEN	
	CONSIDER ANY ILLUMINATED	
	CONSIDER ANY ILLUMINATED	
THROTTLES	OPEN FULL	
	FULL FORWARD	
	CUT-OFF	
*FUEL PUMP	OFF	
*MAG LEFT/RIGHT Switches	ON CLEAR ENGAGE	
*Propeller Area	CLEAR	
*ENG START	ENGAGE	
*MIXTURE	ADVANCE	
*THROTTLE	RETARD	
*OIL PSI	CHECK	
Repeat above procedure (*) for second engine start		
VOLTS	CHECK	
ALTR AMPS		
FUEL PUMPS	OFF	

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ENGINE START WITH EXTERNAL POWER SOURCE

BATT MASTR Switch OFF ALTR LEFT/RIGHTVERIFY OFF

NOTE

The EMERG BATT switch may remain in ARM while using external power. The emergency bus does not receive power from external power.

External Power (24-28 VDC source)CONNECT
Gear Position Indications	3 GREEN
	CONSIDER ANY ILLUMINATED
	CONSIDER ANY ILLUMINATED
THROTTLES	1/4 inch OPEN
PROPELLERS	FULL INCREASE
*FUEL PUMP	ON
*MAG LEFT/RIGHT Switches	ON
*MIXTURE	PRIME - then CUT-OFF
*Propeller Area	CLEAR
*ENG START	ENGAGE
*MIXTURE	ADVANCE
*THROTTLE	ADJUST WHEN ENGINE
	STARTS TO 1000 RPM
*OIL PSI	CHECK

Repeat above procedure (*) for second engine start

THROTTLES	
BATT MASTR Switch	ON
External Power	DISCONNECT
ALTR LEFT/RIGHT Switches	ON
VOLTS	CHECK
ALTR AMP	CHECK
FUEL PUMPS	OFF

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4.5d Warm-Up Checklist

WARM-UP

BEFORE TAXIING

External Power Source VERIFY RI	EMOVED
BATT MASTR SwitchVE	RIFY ON
FUEL SelectorsON, BOTH XFEED (30 :	SEC), ON
AVION MASTER Switch	ON
MFD splash screenVerify Database	Currency
FUEL TotalizerFOB SYNC or ENTER MA	NUALLY
CAS Messages CONSIDER ANY ILLUM	IINATED
PFD Annunciations CONSIDER ANY ILLUM	IINATED
System Messages (Messages Softkey)	. Consider
TRAFFIC (if installed)	
ANN Test Softkey	PRESS

CAUTION

Adjust headset volume to hear the STALL..STALL voice alert at an acceptable level. This will ensure all aural alerts and tones will be audible through the headset.

Standby Flight Instrument	
	or failure annunciations
Altimeters (Standby and PFD)	SET
	AS REQUIRED
CABIN HEAT	AS DESIRED
Radios	CHECK & SET
Autopilot	Verify Preflight Self-Test (PFT)
	completed and disconnect tone heard.
FUEL Selectors	VERIFY ON
Passenger Briefing	COMPLETE
PARK BRAKE	RELEASE
•	

To test the traffic system (if installed), the ADS-B softkey on the Map - Traffic Map page on the MFD must be selected off. The ANN Test softkey is located on the Aux-System Status page of the MFD.

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4.5e Taxiing Checklist

TAXIING

Taxi Area	CLEAR
THROTTLES	APPLY SLOWLY
Brakes	CHECK
Steering	CHECK
Instruments	

NOTE

During taxi, if the VOLTS indication decreases into the warning range, increase engine RPM (if possible) to retain adequate battery charging.

4.5f Ground Check Checklist

GROUND CHECK

PARK BRAKE	SET
MIXTURES	FULL RICH
PROPELLERS	FULL INCREASE
Engine Instruments	CHECK
THROTTLES	1500 RPM
PROPELLERS (Max. Drop - 500 RPM)	FEATHER - CHECK
THROTTLES	2000 RPM
LEFT/RIGHT MAG (Max. Drop - 175 RPM:	
Max. Diff 50 RPM)	CHECK
ALT-AIR	CHECK
THROTTLES	2200 RPM
PROPELLERS (Max. Drop - 300 RPM)	EXERCISE
THROTTLES (550 to 650 RPM)	IDLE - CHECK
FUEL PUMPS	ON
THROTTLES	1000 RPM
Friction Handle	THZ

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4.5f Ground Check Checklist (continued)

If E VOLTS indication less than STARTING ENGINE checklist:	23.3 VOLTS during BEFORE
EMERG BATT Switch	Verify ARM
AVION MASTER Switch	
ALTR LEFT / RIGHT Switches	OFF
BATT MASTR Switch	OFF
E VOLTS Indication	23.3 VOLTS MINIMUM
If E VOLTS less than 23.3 VOLTS, d prior to flight.	
If E VOLTS less than 23.3 VOLTS, d	etermine cause and correct issue
If E VOLTS less than 23.3 VOLTS, deprior to flight. If E VOLTS greater than or equal to	etermine cause and correct issue 23.3 VOLTS:
If E VOLTS less than 23.3 VOLTS, deprior to flight.	etermine cause and correct issue 23.3 VOLTS:ON

Operation of an engine on one magneto should be kept to a minimum.

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4.5f Ground Check Checklist (continued)

The governor can be checked by retarding the propeller control until a drop of 100 RPM to 200 RPM appears, then advancing the throttle to get a slight increase in manifold pressure. The propeller speed should stay the same when the throttle is advanced, indicating proper function of the governor.

Alternate air should also be checked prior to takeoff to be sure the control is operating properly. Avoid prolonged ground operation with alternate air OPEN as the air is unfiltered.

4.5g Before Takeoff Checklist

BEFORE TAKEOFF

Flight Controls	FREE and CORRECT
Flight Instruments	CHECK
Engine Instruments	CHECK
FUEL QTY	SUFFICIENT
PROPELLERS	FULL INCREASE
MIXTURES	FULL RICH
ALT-AIR	CLOSE
COWL FLAPS	OPEN
	CHECK & SET
	SET
	ON
CAS Messages	CONSIDER ANY ILLUMINATED
PFD Annunciations	CONSIDER ANY ILLUMINATED
System Messages (Messages Softkey	/)Consider
Transponder	AS REQUIRED
	ON
PITOT HEAT	AS REQUIRED
	ON
Door	LATCHED (Lower then Upper)
PARK BRAKE	RELEASE

CAUTION

Prior to takeoff with autopilot on, verify that the autopilot servos are disengaged and that flight controls move freely.

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4.5h Takeoff Checklist

CAUTION

Fast taxi turns immediately prior to takeoff should be avoided to prevent unporting fuel feed lines.

NOTE

Adjust mixture prior to takeoff at high elevations. Do not overheat engines. Adjust mixture only enough to obtain smooth engine operation.

NORMAL TAKEOFF

FLAPS	0° to 10°
Stabilator and Rudder Trim	CHECK SET
Power	2700 RPM, FULL THROTTLE
Rotate Speed	75 KIAS
Climb Speed	
GEAR	
FLAPS	UP

NOTE

TAS aural alerts will be muted when GPS altitude is lower than ~ 400 FT AGL.

Takeoff should not be attempted with ice or frost on the wings. Takeoff distances and 50-foot obstacle clearance distances are shown on charts in the Performance Section of this Handbook. The performance shown on charts will be reduced by uphill gradient, tailwind component, soft, wet, rough or grassy surface, or poor pilot technique.

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4.5h Takeoff Checklist (continued)

0° FLAP, SHORT FIELD PERFORMANCE TAKEOFF

FLAPS	UP
Stabilator and Rudder Trim	CHECK SET
Brakes	HOLD
Power	2700 RPM, FULL THROTTLE
MIXTURE	FULL RICH (or SET for ALTITUDE)
Brakes	RELEASE
Rotate Speed	70 KIAS
	82 KIAS
GEAR (with positive rate of climb)UP
Climb Speed (past obstacles)	88 KIAS

NOTE

Gear warning horn will sound when landing gear is retracted with flaps extended beyond first notch.

When a short field effort is required, the safest technique is to use flaps up (0°) . In the event of an engine failure, the airplane will be in the best configuration to maintain altitude immediately after the gear is raised. Refer to Section 5 of this handbook for short field performance data.

4.5i Climb Checklist

MAXIMUM PERFORMANCE CLIMB

Best Rate (Flaps Up)	88 KIAS
Best Angle (Flaps Up)	
COWL FLAPS	
FUEL PUMPS	ON

3

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4.5i Climb Checklist (continued)

CRUISE CLIMB (4.25b)

MIXTURE	FULL RICH
Power	
Climb Speed	105 KIAS
COWL FLAPS	
FUEL PUMPS	

After attaining an altitude for adequate terrain and obstacle clearance, a cruise climb speed of 105 KIAS or higher is recommended. The combination of reduced power and increased climb speed provides better engine cooling, less engine wear, reduced fuel consumption, lower cabin noise level, and better forward visibility. When reducing engine power, the throttles should be retarded first, followed by the propeller controls. Consistent operational use of cruise climb power settings is strongly recommended since this practice will make a substantial contribution to fuel economy, increased engine life, and will reduce the incidence of premature engine overhauls.

4.5j Cruise Checklist

CRUISING

Reference performance charts	and Lycoming Operator's Manual.
Power	SET per Power Setting Chart
FUEL PUMPS	OFF
MIXTURES	ADJUST
COWL FLAPS	As Required
TRIM	As Required

WARNING

Flight in icing conditions is prohibited. If icing is encountered, select alternate air and take immediate action to exit icing conditions. Ensure PITOT HT is ON. Monitor engines and select, ALT-AIR OPEN if roughness or power loss is experienced. Icing is hazardous due to greatly reduced performance, loss of forward visibility, possible longitudinal control difficulties and impaired power plant and fuel system operation.

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4.5j Cruise Checklist (continued)

CRUISING (continued)

Use of the mixture control in cruising flight reduces fuel consumption significantly, especially at higher altitudes. The mixture should be leaned during cruising operation above 5000 ft. altitude and at pilot's discretion at lower altitudes when 75% power or less is being used. If any doubt exists as to the amount of power being used, the mixture should be in the full RICH position for all operations under 5000 feet. To lean the mixture, pull the mixture control aft.

Best economy mixture is obtained by moving the mixture control aft until peak EGT is reached. Performance Cruise mixture is obtained by leaning to peak EGT and then enrichening until the EGT is 125°F rich of the peak value. Under some conditions of altitude and throttle position, the engine may exhibit roughness before peak EGT is reached. If this occurs, the EGT corresponding to the onset of engine roughness should be used as the peak reference value.

For maximum service life, cylinder head temperature should be maintained below 435°F during high performance cruise operation and below 400°F during economy cruise operation. If cylinder head temperatures become too high during flight, reduce them by enriching the mixture, by opening cowl flaps, by reducing power, or by use of any combination of these methods.

The pilot should monitor weather conditions while flying and should be alert to conditions which might lead to icing. If induction system icing is expected, place the ALT-AIR control in the OPEN position.

The LEFT ALTR amd RIGHT ALTR switches should be ON for normal operation. Certain regulator failures can cause the alternator output voltage to increase uncontrollably. To prevent damage, overvoltage relays will automatically shut off the alternator(s). The CAS Warning messages L ALTR FAIL and R ALTR FAIL will warn of this tipped condition.

Alternator outputs will vary with the electrical equipment in use and the state of charge of the battery. ALTR AMPS should not exceed 60 amperes on the ground or 65 amperes in flight. The VOLTS indication will flash red if bus voltage drops below minimum requirements.

IFR operation is not recommended with a single alternator.

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4.5j Cruise Checklist (continued)

CRUISING (continued)

Since the Seminole has one fuel tank per engine, it is advisable to feed the engines symmetrically during cruise so that the same approximate amount of fuel will be left in each side for the landing. The crossfeed (XFEED) can be used to balance FUEL QTY, if necessary.

During flight, keep account of time and fuel used in connection with power settings to verify the accuracy of the fuel flow and fuel quantity gauging systems.

There are no mechanical uplocks in the landing gear system. If the hydraulic system malfunctions, the landing gear will free-fall to the gear down position. True airspeed with gear down is approximately 75% of the gear retracted airspeed for any given power setting. Allowances for the reduction in airspeed and range should be made when planning extended flight between remote airfields or flight over water.

4.5k Descent Checklist

DESCENT

MIXTURE	ADJUST with Descent
THROTTLES	As Required
COWL FLAPS	^

When power is reduced for descent, the mixtures should be enriched as altitude decreases. The propellers may be left at cruise setting; however, if the propeller speed is reduced, it should be done after the throttles have been retarded. Cowl flaps should normally be closed to keep the engines at the proper operating temperature.

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4.51 Approach and Landing Checklist

APPROACH AND LANDING

Seat Backs	ERECT
Seat Belts, Harnesses	ADJUSTED
FUEL PUMPS	ON
FUEL Selectors	ON
GEAR (Below 140 KIAS)	
Gear Position Indications	3 GREEN
Nacelle Mirror	CHECK NOSE GEAR DOWN
MIXTURES	FULL RICH
PROPELLERS	FULL INCREASE
ALT-AIR	AS REQUIRED
Autopilot	Disconnect (above 200 FT AGL)

NOTE

TAS aural alerts will be muted when GPS altitude is lower than ~ 400 FT AGL.

NOTE

The HSI will auto-slew during CDI transitions to LOC, LOC BC, LDA, or SDF approaches if the approach is activated in the G1000 system. The pilot should always double check the inbound course pointer prior to initiating a VHF NAV approach.

During the approach for a landing, the CHECK GEAR aural alert may sound. The mutable CHECK GEAR is triggered when the gear is not down and locked and manifold pressure(s) is less than 14 in Hg. The non-mutable CHECK GEAR is triggered when the landing gear is not down and locked and flaps are extended beyond the first notch. The severity of the CHECK GEAR CAS message is determined by proximity to the ground. A Caution message is triggered when above ~400 feet AGL, and a Warning is triggered below ~400 feet AGL. See Section 7 for additional details.

The landing gear is down and locked when three solid green circles are indicated on the MFD. The mirror on the left engine nacelle may be used to visually verify the nose landing gear position.

Operate the toe brakes to verify sufficient pressure for normal braking. Verify the parking brake is not set.

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4.51 Approach and Landing Checklist (continued)

NORMAL LANDING

FLAPS	0° to FULL DOWN
Airspeed (Flaps Up)	80-90 KIAS
(Flaps Down)	
Trim	AS REQUIRED
THROTTLES	AS REQUIRED
Touchdown	-
Braking	AS REQUIRED
•	~ · · · · · · · · · · · · · · · · · · ·

Landing may be made with any flap setting. Normally full flaps are used, which reduces stall speed and permits slower speed at contact. Maximum braking after touch-down is achieved by retracting the flaps, applying back pressure to the wheel and applying pressure on the brakes. However, unless maximum braking is needed or unless a strong crosswind or gusty condition exists, it is best to wait until turning off the runway to retract the flaps. This will permit full attention to be given to the landing and landing rollout.

If a crosswind or high-wind landing is necessary, approach with higher than normal speed and with zero to 25 degrees of flaps. Immediately after touch-down, raise the flaps. During a crosswind approach hold a crab angle into the wind until ready to flare out for the landing. Then lower the wing that is into the wind to eliminate the crab angle without drifting, and use the rudder to keep the wheels aligned with the runway. Avoid prolonged side slips with a low fuel indication.

NOTE

The maximum demonstrated crosswind component during landing is 17 KTS.

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4.51 Approach and Landing Checklist (continued)

SHORT FIELD PERFORMANCE LANDING

FLAPS (Below 111 KIAS)	FULL DOWN
	75 KIAS
Trim	AS REQUIRED
THROTTLES	IDLE
Touchdown	MAIN WHEELS
Braking	MAXIMUM without SKIDDING

For landings on short runways or runways with adjacent obstructions, a short field landing technique should be used in accordance with the charts in Section 5. The airplane should be flown down final with full flaps, at 75 KIAS (at maximum weight) so as to cross any obstructions with the throttles at idle. Immediately after touch-down, raise the flaps and apply back pressure to the control wheel as maximum braking is applied.

4.5m Go-Around Checklist

GO-AROUND

MIXTURES	FULL RICH
PROPELLERS	FULL INCREASE
THROTTLES	FULL OPEN
Control Wheel	BACK PRESSURE TO OBTAIN
	POSITIVE CLIMB ATTITUDE
	RETRACT INCREMENTALLY
GEAR	UP
	AS REQUIRED

WARNING.

Autopilot coupled go-around is not authorized during single engine operations.

If the aircraft is equipped with optional Underspeed Protection (USP) and an autopilot coupled go-around is desired, press the TO/GA button on the throttle handle, followed immediately by the checklist shown above. Refer to Section 7 for additional details on the autopilot coupled go-around.

4.5n After Landing Checklist

AFTER LANDING

Clear of runway

Cical of fullway.	
FLAPS	RETRACT
COWL FLAPS	FULL OPEN
ALT-AIR	CLOSE
FUEL PUMPS	OFF
LIGHTS	AS REOUIRED
PITOT HEAT	

A spongy pedal during braking, is often an indication that the brake fluid needs replenishing.

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4.50 Stopping Engine Checklist

STOPPING ENGINE

CABIN HEAT (If ON)	FAN - 2 MIN. THEN OFF
VENT FAN	OFF
AVION MASTER	OFF
EMERG BATT	
LEFT/RIGHT ALTR	OFF
LEFT/RIGHT FUEL PUMP	
All Other Electrical Equipment	OFF
THROTTLES	
MIXTURES	
LEFT/RIGHT ENG MAG Switches	OFF
Interior Lights (at night)	OFF
Exterior Lights	
BATT MASTR	
STANDBY INSTRUMENT	

NOTE

The flaps must be placed in the "UP" position for the flap step to support weight. Passengers should be cautioned accordingly.

NOTE

In case the standby instrument remains "ON" due to improper shutdown, the unit switches to internal battery and depletes it. To turn off the Aspen EBD, press the "SHUT DOWN" command from Main Menu page 6 or hold the red "REV" button for 20 seconds. To turn off the Garmin G5, press and hold the power button for five seconds.

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4.5p Mooring Checklist

MOORING

PARK BRAKE	AS REQUIRED
Control Wheel	SECURED with seat belts
FLAPS	FULL UP
Wheel Chocks	IN PLACE
Tiedowns	

If necessary, the airplane should be moved on the ground with the aid of the optional nose wheel tow bar.

The ailerons and stabilator should be secured by looping the seat belt through the control wheel and pulling it snug. The rudder need not be secured under normal conditions, as its connection to the nose wheel holds it in position. The flaps are locked when in the fully retracted position. Wheel chocks should be positioned in place, or the parking brake set. Tie-down ropes may be attached to mooring rings under each wing and to the tail skid.

4.5q VSSE - Intentional One Engine Inoperative Speed

VSSE is not a limitation. However, it is recommended that, except for training, demonstrations, takeoffs, and landings, the airplane should not be flown at a speed slower than VSSE.

The intentional one engine inoperative speed, Vsse, for the PA-44-180 is 82 KIAS.

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4.5r VMCA - Air Minimum Control Speed

The VMCA demonstration, may be required for the FAA flight test for the multi-engine rating.

VMCA DEMONSTRATION

(a) Landing Gear	UP
(b) Flaps	UP
(c) Altitude	at or above 4000 ft. AGL
(d) Airspeed	at or above 82 KIAS (Vsse)
(e) Mixture	FULL RICH
(f) Propeller Controls	HIGH RPM
(g) Throttle (Simulated Inope	erative Engine)IDLE
(h) Throttle (Other Engine)	FULL FORWARD
(i) Airspeed	Reduce approximately 1 knot per second
:	•
	CONTROL TRAVEL or VMCA is obtained

CAUTION

Use rudder to maintain directional control (heading) and ailerons to maintain 5° bank towards the operative engine (lateral attitude). At the first sign of either VMCA (airspeed indicator redline) or stall warning (which may be evidenced by: inability to maintain heading or bank attitude, aerodynamic stall buffet, or stall warning horn), immediately initiate recovery; reduce power to idle on the operative engine, and immediately lower the nose to regain VMCA and continue accelerating to VSSE.

CAUTION

One engine inoperative stalls are not recommended.

Under no circumstances should an attempt be made to fly at a speed below VMCA with only one engine operating.

4.5s Practice One Engine Inoperative Flight

Simulated one engine inoperative flight can be practiced without actually shutting down one engine by setting the propeller rpm of an engine to approximate zero thrust. This is accomplished at typical training altitudes with the throttle adjusted to produce the appropriate engine speed shown below and the mixture full rich, or leaned as required for smooth low power operation.

CAUTION

A rapid reduction in power (full throttle to idle in less than 2 seconds) may be harmful to the engine.

Propeller rpm for Zero Thrust	
KIAS	RPM
82 Vsse	1850
88 Vyse	2180
100	2510
110	2690

4.5t Noise Level

The corrected noise levels for this aircraft are as follows:

Propeller Designation	14 CFR Part 36	ICAO Annex 16
Straight Blade HC-C2Y(K,R)-2CEUF/FC7666A-2R HC-C2Y(K,R)-2CLEUF/FJC7666A-2R	74.7 dB(A) (Amdt. 36-4)	74.7 dB(A) (Vol I, Part II, Ch 6)
Scimitar Blade HC-C2YR-2CEUFP/FC7497 HC-C2YR-2CLEUFP/FJC7497	82.4 dB(A) (Amdt. 36-28)	82.4 dB(A) (Vol I, Ch 10)

No determination has been made by the Federal Aviation Administration that the noise levels of this airplane are or should be acceptable or unacceptable for operation at, into, or out of any airport.

The above statement notwithstanding, the noise level stated above has been verified by and approved by the Federal Aviation Administration in noise level test flights conducted in accordance with 14 CFR Part 36 - Noise Standards: Aircraft Type and Airworthiness Certification. This aircraft model is in compliance with all 14 CFR Part 36 noise standards applicable to this type.

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4.5u Stalls

The loss of altitude during a power off stall with the gear and flaps retracted may be as much as 300 feet.

NOTE

The stall warning system is inoperative with the Battery Master OFF

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